What is claimed is:

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- 1. A method of estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance, wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said method receiving, for a plurality of two or more time moments j, the following inputs:
- a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,
- a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,
- a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),
- a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),
- a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),
- a vector $\boldsymbol{\varphi_j}^{R}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),
- a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, said method comprising the steps of:
- (a) generating, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R \gamma_j^B) (D_j^R D_j^B);$

(b) generating, for each time moment j, a vector Δφ_j of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in
 30 the form of:

$$\Delta \varphi_j = (\varphi_j^R - \varphi_j^B) - \Lambda^{-1} \cdot (D_j^R - D_j^B),$$

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where Λ^{-1} is a diagonal matrix comprising the inverse wavelengths of the satellites;

- (c) generating, for time moment j = 1, an LU-factorization of a matrix \mathbf{M}_1 or a matrix inverse of matrix \mathbf{M}_1 , the matrix \mathbf{M}_1 being a function of at least $\boldsymbol{\Lambda}^{-1}$ and $\boldsymbol{H}_1^{\gamma}$;
- (d) generating, for time moment j = 1, a vector \mathbf{N}_1 as a function of at least $\Delta \gamma_I$, $\Delta \varphi_I$, and the LU-factorization of matrix \mathbf{M}_1 or the matrix inverse of matrix \mathbf{M}_1 ;
- (e) generating, for an additional time moment $j \neq 1$, an LU-factorization of a matrix \mathbf{M}_j or a matrix inverse of matrix \mathbf{M}_j , the matrix \mathbf{M}_j being a function of at least Λ^{-1} , \mathbf{H}_j^{γ} and an instance of matrix \mathbf{M} generated for a different time moment; and
- (f) generating, for an additional time moment $j \neq 1$, a vector N_j as a function of at least $\Delta \gamma_j$, $\Delta \varphi_j$, and the LU-factorization or matrix M_j or the matrix inverse of matrix M_j , the vector N_j having estimates of the floating ambiguities.
- 2. The method of Claim 1 wherein step (c) comprises generating an LU-factorization for a matrix comprising a form equivalent to (G^TP_1G) , where:
- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix \mathbf{P}_1 has 2n rows, 2n columns, and a form which comprises a matrix equivalent to $\left(\mathbf{R}_1^{-1} \mathbf{R}_1^{-1} \mathbf{Q}_1 \left(\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1\right)^{-1} \mathbf{Q}_1^T \mathbf{R}_1^{-1}\right)$ where the matrix \mathbf{R}_1 is a weighting matrix, where the matrix \mathbf{R}_1^{-1} comprises an inverse of matrix \mathbf{R}_1 , where the matrix \mathbf{Q}_1 has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of \mathbf{Q}_1 comprising matrix \mathbf{H}_1^{γ} and the other of the sub-matrices of \mathbf{Q}_1 comprising the matrix product $\mathbf{\Lambda}^{-1}\mathbf{H}_1^{\gamma}$, and wherein the matrix \mathbf{Q}_1^T comprises the transpose of matrix \mathbf{Q}_1 , and where the quantity $\mathbf{q}\mathbf{S}_k$ is a zero matrix when the distance

between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

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- 3. The method of Claim 2 wherein step (d) comprises the step of generating vector \mathbf{N}_1 to comprise a vector having a form equivalent to $\mathbf{M}_1^{-1} \left(\mathbf{G}^T \mathbf{P}_1 \ \boldsymbol{\mu}_1 + q \mathbf{g}_1 \right)$, where the matrix \mathbf{M}_1^{-1} comprises an inverse of matrix of matrix \mathbf{M}_1 , and where the vector $\boldsymbol{\mu}_1$ comprises the vector $[\Delta \boldsymbol{\gamma}_1, \Delta \boldsymbol{\varphi}_1]^T$, and where the quantity $q \mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.
- 4. The method of Claim 1 wherein step (e) comprises generating an LU-factorization for a matrix comprising a form equivalent to $\mathbf{M}_j = \mathbf{M}_{j-1} + \mathbf{G}^T \mathbf{P}_j \mathbf{G}$, where:
- \mathbf{M}_{j-1} comprises the matrix \mathbf{M}_1 of step (c) when j = 2 and comprises the matrix \mathbf{M}_j of step (e) for the j-1 time moment when j > 2,
- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprising an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix P_j has 2n rows, 2n columns, and a form which comprise a matrix equivalent to (R_j⁻¹ R_j⁻¹ Q_j(Q_j^T R_j⁻¹ Q_j + qS_j)⁻¹Q_j^T R_j⁻¹) where the matrix R_j is a weighting matrix, where the matrix R_j⁻¹ comprises an inverse of matrix R_j, where the matrix Q_j has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_j comprising matrix H_j^γ and the other of the sub-matrices of Q_j
 comprising the matrix product Λ⁻¹H_j^γ, and wherein the matrix Q_j^T comprises the transpose of matrix Q_j, and where the quantity qS_j is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may

be a non-zero weighting parameter and S_j may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

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5. The method of Claim 4 wherein step (f) comprises the step of generating vector N_j to comprise a vector having a form equivalent to

 $\mathbf{N}_{j-1} + \mathbf{M}_{j}^{-1} \left[\mathbf{G}^{T} \mathbf{P}_{j} \left(\mathbf{\mu}_{j} - \mathbf{G} \mathbf{N}_{j-1} \right) + q \mathbf{g}_{j} \right]$, where the matrix \mathbf{M}_{j}^{-1} comprises an inverse of matrix of matrix \mathbf{M}_{j} , where the vector $\boldsymbol{\mu}_{j}$ comprises the vector $\left[\Delta \boldsymbol{\gamma}_{j}, \Delta \boldsymbol{\varphi}_{j} \right]^{T}$, and where the vector \mathbf{N}_{j-1} comprises the vector \mathbf{N}_{1} generated by step (d) when j = 2 and comprises the vector \mathbf{N}_{j-1} generated by step (f) for the j-1 time moment when j > 2, and where the quantity $q\mathbf{g}_{j}$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_{j} may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

6. The method of claim 3 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) generates matrix S_1 in a form equivalent to:

$$\mathbf{S}_{1} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{1}\|} \mathbf{r}_{1} \mathbf{r}_{1}^{T}$$

where \mathbf{r}_1 is a vector comprising estimates of the three coordinates of the baseline vector for the time moment j=1 and a zero as fourth component, where \mathbf{r}_1^T is the vector transpose of \mathbf{r}_1 , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_1 for the time moment j=1 in a form equivalent to:

$$\mathbf{g}_1 = \mathbf{G}^T \mathbf{R}_1^{-1} \mathbf{Q}_1 (\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1)^{-1} \mathbf{h}_1,$$

where:

$$\mathbf{h}_1 = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_1\|}\right) \mathbf{r}_1.$$

- 7. The method of Claim 3 wherein weighting matrix \mathbf{R}_1 comprises an identity matrix multiplied by a scalar quantity.
- 8. The method of claim 5 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (e) generates matrix S_j in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right)\left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} \quad 0\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\mathbf{r}j\mathbf{r}j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (f) generates vector \mathbf{g}_{j} for the j-th time moment in a form equivalent to:

$$\mathbf{g}_{j} = \mathbf{G}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} (\mathbf{Q}_{j}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} + q \mathbf{S}_{j})^{-1} \mathbf{h}_{j},$$

where:

$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

- 9. The method of Claim 5 wherein the weighting matrix \mathbf{R}_j comprises an identity matrix multiplied by a scalar quantity for at least one time moment j.
- 10. The method of Claim 4 wherein the generation of the LU-factorization in step (e) comprises the steps of:
- (g) generating an LU-factorization of matrix \mathbf{M}_{j-1} in a form equivalent to $\mathbf{L}_{j-1} \mathbf{L}_{j-1}^T$ wherein \mathbf{L}_{j-1} is a low-triangular matrix and \mathbf{L}_{j-1}^T is the transpose of \mathbf{L}_{j-1} ;
- (h) generating a factorization of $\mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$ in a form equivalent to $\mathbf{T}_{j} \mathbf{T}_{j}^{T} = \mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$, where \mathbf{T}_{j}^{T} is the transpose of \mathbf{T}_{j} ;

- (i) generating an LU-factorization of matrix \mathbf{M}_j in a form equivalent to $\mathbf{L}_j \mathbf{L}_j^T$ from a plurality n of rank-one modifications of matrix \mathbf{L}_{j-1} , each rank-one modification being based on a respective column of matrix \mathbf{T}_j , where n is the number of rows in matrix \mathbf{M}_j .
- 11. The method of Claim 10 wherein step (h) generates matrix \mathbf{T}_j from a Cholesky factorization of $\mathbf{G}^{\mathsf{T}} \mathbf{P}_j \mathbf{G}$.
- 12. The method of Claim 10 wherein weighting matrix \mathbf{R}_j has a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein $\mathbf{R}^{oldsymbol{\gamma}}$ and $\mathbf{R}^{oldsymbol{arphi}}$ are related to a common weighting matrix \mathbf{W} and scaling

5 parameters σ_{γ} and σ_{φ} as follows $\left(\mathbf{R}^{\gamma}\right)^{-1} = \frac{1}{\sigma_{\gamma}^{2}}\mathbf{W}$, and $\left(\mathbf{R}^{\varphi}\right)^{-1} = \frac{1}{\sigma_{\varphi}^{2}}\mathbf{W}$,

wherein step (h) of generating matrix T_j comprises the steps of:

generating a scalar b in a form equivalent to: $b = \frac{\sigma_{\gamma}^2 \lambda_{GPS}^2}{\sigma_{\gamma}^2 + \lambda_{GPS}^2 \sigma_{\varphi}^2}$, where λ_{GPS} is

the wavelength of the satellite signals,

generating a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_{j}^{\gamma}$,

generating a Householder matrix \mathbf{S}_{HH} for matrix $\widetilde{\mathbf{H}}$, and generating matrix \mathbf{T}_j in a form equivalent to:

$$\mathbf{T}_{j} = \frac{1}{\sigma_{\varphi}} \mathbf{W}^{1/2} \mathbf{S}_{\mathbf{HH}} \begin{bmatrix} \sqrt{\left(1 - \frac{b}{\lambda_{GPS}^{2}}\right)} \mathbf{I}_{4} & | & \mathbf{O}_{4 \times (n-4)} \\ & --- & | & ---- \\ \mathbf{O}_{(n-4) \times 4} & | & \mathbf{I}_{(n-4) \times (n-4)} \end{bmatrix}.$$

13. The method of Claim 10 wherein weighting matrix \mathbf{R}_j is applied to a case where there is a first group of satellite signals having carrier signals in a first wavelength

band and a second group of satellite signals having a carrier signals in a second wavelength band, the weighting frequency having a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein \mathbf{R}^{γ} and \mathbf{R}^{φ} are related to a common weighting matrix \mathbf{W} , the carrier wavelengths of the first group of signals as represented by matrix $\Lambda^{(1)}$, the carrier wavelengths of the second group of signals as represented by matrix $\Lambda^{(2)}$, the center wavelength of the first band as represented by λ_1 , the center wavelength of the first band as represented by λ_2 , and scaling parameters σ_{γ} and σ_{φ} , as follows:

$$\left(\mathbf{R}_{\mathbf{j}}^{\gamma}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} \end{bmatrix},$$

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$$\left(\mathbf{R}_{\mathbf{j}}^{\varphi}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(1)} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(2)} \end{bmatrix},$$

where
$$W^{(1)} = \frac{1}{\lambda_1^2} \Lambda^{(1)} W \Lambda^{(1)}$$
, $W^{(2)} = \frac{1}{\lambda_2^2} \Lambda^{(2)} W \Lambda^{(2)}$,

wherein step (h) of generating matrix T_i comprises the steps of:

generating a scalar
$$b$$
 in a form equivalent to:
$$b = \frac{\sigma_r^2 \lambda_1^2 \lambda_2^2}{2\sigma_\varphi^2 \lambda_1^2 \lambda_2^2 + \sigma_r^2 \lambda_1^2 + \sigma_r^2 \lambda_2^2},$$

where λ_1 is the wavelength of a first group of satellite signals and λ_2 is the wavelength of a second group of satellite signals,

generating a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_{i}^{\gamma}$,

generating a Householder matrix S_{HH} for matrix \tilde{H} , and

generating matrix
$$\mathbf{T}_j$$
 in a form equivalent to: $\mathbf{T}_j = \frac{1}{\sigma_{\varphi}} \begin{bmatrix} \mathbf{A11} & | & \mathbf{O_{n \times n}} \\ --- & | & --- \\ \mathbf{A21} & | & \mathbf{A22} \end{bmatrix}$

where sub-matrixces A11, A21, and A22 are as follows:

A11 =
$$(W^{(1)})^{\frac{1}{2}}S_{HH}\begin{bmatrix} \sqrt{1-b/\lambda_{l}^{2}} I_{4} & | & O_{4\times(n-4)} \\ ----- & | & ----- \\ O_{(n-4)\times4} & | & I_{(n-4)} \end{bmatrix}$$
,

A21 =
$$\left(\mathbf{W^{(2)}}\right)^{\frac{1}{2}} \mathbf{S_{HH}} \begin{bmatrix} -\frac{b}{\lambda_1 \lambda_2 \sqrt{1-b/\lambda_1^2}} \mathbf{I_4} & | & \mathbf{O_{4\times(n-4)}} \\ ----- & | & ----- \\ \mathbf{O_{(n-4)\times 4}} & | & \mathbf{O_{(n-4)\times(n-4)}} \end{bmatrix}$$
, and

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$$\mathbf{A22} = \left(\mathbf{W^{(2)}}\right)^{\frac{1}{2}} \mathbf{S_{HH}} \begin{bmatrix} \sqrt{\frac{1 - b/\lambda_1^2 - b/\lambda_2^2}{1 - b/\lambda_1^2}} \mathbf{I_4} & | & \mathbf{O_{4 \times (n-4)}} \\ ----- & | & ----- \\ \mathbf{O_{(n-4) \times 4}} & | & \mathbf{I_{(n-4)}} \end{bmatrix}.$$

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14. The method of Claim 4 wherein step (d) comprises generating a vector \mathbf{B}_1 to comprise a vector having a form equivalent to $\mathbf{G}^T \mathbf{P}_1 \mu_1 + q \mathbf{g}_1$, where the vector μ_1 comprises the vector $[\Delta \gamma_1, \Delta \varphi_1]^T$, and where the quantity $q \mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_j may be a non-zero vector when the distance between the first and second navigation receivers is constrained; and

wherein step (f) further comprises generating, for each time moment $j \neq 1$, a vector \mathbf{B}_j to comprise a matrix having a form equivalent to $\mathbf{B}_{j-1} + \mathbf{G}^T \mathbf{P}_j \mu_j + q \mathbf{g}_j$, where the vector μ_j comprises the vector $[\Delta \gamma_j, \Delta \varphi_j]^T$, and where the vector \mathbf{B}_{j-1} is the vector \mathbf{B}_1 generated by step (d) when j = 2 and comprises the vector generated by step (f) for the for the j-1 time moment when j > 2, and where the quantity $q \mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be

non-zero and \mathbf{g}_{i} may be a non-zero vector when the distance between the first and second navigation receivers is constrained; and

wherein step (f) further comprises generating vector \mathbf{N}_j to comprise a vector having a form equivalent to $\mathbf{N}_j = [\mathbf{M}_j]^{-1} \mathbf{B}_j$, where the matrix \mathbf{M}_j^{-1} comprises an inverse of matrix of matrix \mathbf{M}_j .

15. The method of claim 14 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) generates matrix S_1 in a form equivalent to:

$$\mathbf{S}_{1} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{1}\|} \mathbf{r}_{1} \mathbf{r}_{1}^{T}$$

where \mathbf{r}_1 is a vector comprising estimates of the three coordinates of the baseline vector for the time moment j=1 and a zero as fourth component, where \mathbf{r}_1^T is the vector transpose of \mathbf{r}_1 , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_1 for the time moment j=1 in a form equivalent

$$\mathbf{g}_1 = \mathbf{G}^T \mathbf{R}_1^{-1} \mathbf{Q}_1 (\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1)^{-1} \mathbf{h}_1,$$

where:

to:

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$$\mathbf{h}_1 = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_1\|}\right) \mathbf{r}_1.$$

16. The method of claim 14 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (e) generates matrix S_j in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right)\left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1}\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\mathbf{r}j\mathbf{r}j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector

transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (f) generates vector \mathbf{g}_{j} for the j-th time moment in a form equivalent to:

$$\mathbf{g}_{j} = \mathbf{G}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} (\mathbf{Q}_{j}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} + q \mathbf{S}_{j})^{-1} \mathbf{h}_{j},$$

where:

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$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

- 17. The method of Claim 14 wherein the weighting matrix \mathbf{R}_j comprises an identity matrix multiplied by a scalar quantity for at least one time moment j.
- 18. A method of estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R), wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said method receiving, for a plurality of two or more time moments j, the following inputs for each time moment j:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

- a vector $\boldsymbol{\varphi_j}^{R}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),
 - a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, said method comprising the steps of:
- (a) generating, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R \gamma_j^B) (D_j^R D_j^B)$, said step generating a set of range residuals $\Delta \gamma_k$, k=1,...,j;
- (b) generating, for each time moment j, a vector $\Delta \varphi_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \varphi_j = (\varphi_j^R \varphi_j^B) \Lambda^{-1} \cdot (D_j^R D_j^B)$, where Λ^{-1} is a diagonal matrix comprising the inverse wavelengths of the satellites, said step generating a set of phase residuals $\Delta \varphi_k$, k=1,...,j;
 - (c) generating an LU-factorization of a matrix \mathbf{M} or a matrix inverse of matrix \mathbf{M} , the matrix \mathbf{M} being a function of at least $\mathbf{\Lambda}^{-1}$ and \mathbf{H}_k^{γ} , for index k of \mathbf{H}_k^{γ} covering at least two of the time moments j;
 - (d) generating a vector N of estimated floating ambiguities as a function of at least the set of range residuals $\Delta \gamma_k$, the set of phase residuals $\Delta \varphi_k$, and the LU-factorization of matrix M or the matrix inverse of matrix M.
 - 19. The method of Claim 18 wherein step (c) comprises generating matrix \mathbf{M} in a form equivalent to the summation $\left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{G})\right]$,

where:

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- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and

- the matrix P_k has 2n rows, 2n columns, and a form which comprises a matrix equivalent to P_k = R_k⁻¹ R_k⁻¹ Q_k (Q_k^T R_k⁻¹ Q_k + qS_k)⁻¹ Q_k^T R_k⁻¹, where the matrix R_k is a weighting matrix, where the matrix R_k⁻¹ comprises an inverse of matrix R_k, where the matrix Q_k has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_k comprising matrix H_k^γ and the other of the sub-matrices of Q_k comprising the matrix product Λ⁻¹H_k^γ, and wherein the matrix Q_k^T comprises the transpose of matrix Q_k, and where the quantity qS_k is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.
 - 20. The method of Claim 19 wherein step (d) comprises generating matrix N in a form equivalent to:

$$\mathbf{N} = \mathbf{M}^{-1} \times \left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{\mu}_{k} + q \mathbf{g}_{k}) \right],$$

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where the matrix \mathbf{M}^{-1} comprises an inverse of matrix of matrix \mathbf{M} , where the vector $\boldsymbol{\mu}_k$ comprises the vector $[\boldsymbol{\Delta}\gamma_k, \boldsymbol{\Delta}\varphi_k]^T$, and where the quantity $q\mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

21. The method of claim 20 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) generates matrix S_k for the k-th time moment in a form equivalent to:

$$\mathbf{S}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{k}\|} \mathbf{r}_{k} \mathbf{r}_{k}^{T}$$

where \mathbf{r}_k is a vector comprising estimates of the three coordinates of the baseline vector at the k-th time moment, and a zero as fourth component, where \mathbf{r}_k^T is the vector transpose of \mathbf{r}_k , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_k for the k-th time moment in a form equivalent

$$\mathbf{g}_k = \mathbf{G}^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k (\mathbf{Q}_k^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k + q \mathbf{S}_k)^{-1} \mathbf{h}_k,$$

where:

to:

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$$\mathbf{h}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \mathbf{r}_{k}.$$

- 22. The method of Claim 19 wherein at least one of the weighting matrices \mathbf{R}_k comprises an identity matrix multiplied by a scalar quantity.
- 23. A computer program product for directing a data processor to estimate a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance, wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, the process receiving, for a plurality of two or more time moments j, the following inputs:

a vector \mathbf{y}_{j}^{B} representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{B}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

a vector $\boldsymbol{\varphi_j}^R$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),

a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, the computer program product comprising:

a computer-readable medium;

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a first set of instructions embodied on the computer-readable medium which directs the data processor to generate, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R - \gamma_j^B) - (D_j^R - D_j^B)$;

a second set of instructions embodied on the computer-readable medium which directs the data processor to generate, for each time moment j, a vector $\Delta \boldsymbol{\varphi}_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \boldsymbol{\varphi}_j = (\boldsymbol{\varphi}_j^R - \boldsymbol{\varphi}_j^B) - \boldsymbol{\Lambda}^{-1} \cdot (\boldsymbol{D}_j^R - \boldsymbol{D}_j^B)$, where $\boldsymbol{\Lambda}^{-1}$ is a diagonal matrix comprising the inverse wavelengths of the satellites;

a third set of instructions embodied on the computer-readable medium which directs the data processor to generate, for time moment j = 1, an LU-factorization of a matrix \mathbf{M}_1 or a matrix inverse of matrix \mathbf{M}_1 , the matrix \mathbf{M}_1 being a function of at least $\mathbf{\Lambda}^{-1}$ and \mathbf{H}_1^{γ} ;

a fourth set of instructions embodied on the computer-readable medium which directs the data processor to generate, for time moment j = 1, a vector \mathbf{N}_1 as a function of at least $\Delta \gamma_1$, $\Delta \varphi_1$, and the LU-factorization of matrix \mathbf{M}_1 or the matrix inverse of matrix \mathbf{M}_1 ;

a fifth set of instructions embodied on the computer-readable medium which directs the data processor to generate, for an additional time moment $j \neq 1$, an LU-factorization of a matrix \mathbf{M}_j or a matrix inverse of matrix \mathbf{M}_j , the matrix \mathbf{M}_j being a function of at least Λ^{-1} and \mathbf{H}_j^{γ} ; and

a sixth set of instructions embodied on the computer-readable medium which directs the data processor to generate, for an additional time moment $i \neq 1$, a vector N_i as a

- function of at least $\Delta \gamma_j$, $\Delta \varphi_j$, and the LU-factorization or matrix M_j or the matrix inverse of matrix M_j , the vector N_j having estimates of the floating ambiguities.
 - 24. The computer program product of Claim 23 wherein the third set of instructions directs the data processor to generate an LU-factorization for a matrix comprising a form equivalent to $(\mathbf{G}^{T}\mathbf{P}_{1}\mathbf{G})$, where:
 - the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and

- the matrix \mathbf{P}_1 has 2n rows, 2n columns, and a form which comprises a matrix equivalent to $\left(\mathbf{R}_1^{-1} \mathbf{R}_1^{-1} \mathbf{Q}_1 \left(\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1\right)^{-1} \mathbf{Q}_1^T \mathbf{R}_1^{-1}\right)$ where the matrix \mathbf{R}_1
- is a weighting matrix, where the matrix \mathbf{R}_1^{-1} comprises an inverse of matrix \mathbf{R}_1 , where the matrix \mathbf{Q}_1 has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of \mathbf{Q}_1 comprising matrix \mathbf{H}_1^{γ} and the other of the sub-matrices of \mathbf{Q}_1 comprising the matrix product $\mathbf{\Lambda}^{-1}\mathbf{H}_1^{\gamma}$, and wherein the matrix $\mathbf{Q}_1^{\mathrm{T}}$ comprises the transpose of matrix \mathbf{Q}_1 , and where the quantity $\mathbf{q}\mathbf{S}_k$ is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where \mathbf{q} may be a non-zero weighting parameter and \mathbf{S}_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.
 - 25. The computer program product of Claim 24 wherein the fourth set of instructions directs the data processor to generate vector \mathbf{N}_1 to comprise a vector having a form equivalent to $\mathbf{M}_1^{-1} \left(\mathbf{G}^T \mathbf{P}_1 \ \boldsymbol{\mu}_1 + q \mathbf{g}_1 \right)$, where the matrix \mathbf{M}_1^{-1} comprises an inverse of matrix of matrix \mathbf{M}_1 , and where the vector $\boldsymbol{\mu}_1$ comprises the vector $[\Delta \boldsymbol{\gamma}_1, \Delta \boldsymbol{\varphi}_1]^T$, and where the quantity $q \mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

- 26. The computer program product of Claim 23 wherein the fifth set of instructions directs the data processor to generate an LU-factorization for a matrix comprising a form equivalent to $\mathbf{M}_{j} = \mathbf{M}_{j-1} + \mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$, where:
- \mathbf{M}_{j-1} comprises the matrix \mathbf{M}_1 of step (c) when j = 2 and comprises the matrix \mathbf{M}_j of step (e) for the j-1 time moment when j > 2,
- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprising an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and

- the matrix P_j has 2n rows, 2n columns, and a form which comprise a matrix equivalent to (R_j⁻¹ R_j⁻¹ Q_j(Q_j^T R_j⁻¹ Q_j + qS_j)⁻¹Q_j^T R_j⁻¹) where the matrix R_j is a weighting matrix, where the matrix R_j⁻¹ comprises an inverse of matrix R_j, where the matrix Q_j has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_j comprising matrix H_j^γ and the other of the sub-matrices of Q_j
 comprising the matrix product Λ⁻¹H_j^γ, and wherein the matrix Q_j^T comprises the transpose of matrix Q_j, and where the quantity qS_j is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_j may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.
 - 27. The computer program product of Claim 26 wherein the sixth set of instructions directs the data processor to generate a vector \mathbf{N}_j to comprise a vector having a form equivalent to $\mathbf{N}_{j-1} + \mathbf{M}_j^{-1} \left[\mathbf{G}^T \mathbf{P}_j \left(\mathbf{\mu}_j \mathbf{G} \mathbf{N}_{j-1} \right) + q \mathbf{g}_j \right]$, where the matrix \mathbf{M}_j^{-1} comprises an inverse of matrix of matrix \mathbf{M}_j , where the vector $\mathbf{\mu}_j$ comprises the vector $[\Delta \gamma_j, \Delta \varphi_j]^T$, and where the vector \mathbf{N}_{j-1} comprises the vector \mathbf{N}_1 generated by step (d) when j = 2 and comprises the vector \mathbf{N}_{j-1} generated by step (f) for the j-1 time moment when j > 2, and where the quantity $q\mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and

 \mathbf{g}_{j} may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

28. The computer program product of Claim 27 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , and wherein the fifth set of instructions directs the data processor to generate matrix S_j in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right)\left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} \quad 0\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\mathbf{r}j\mathbf{r}j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein the sixth set of instructions directs the data processor to generate vector \mathbf{g}_j for the j-th time moment in a form equivalent to:

$$\mathbf{g}_{j} = \mathbf{G}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} (\mathbf{Q}_{j}^{T} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} + q \mathbf{S}_{j})^{-1} \mathbf{h}_{j},$$

where:

$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

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29. The computer program product of Claim 28 wherein the fifth set of instructions comprises:

a seventh set of instructions that direct the data processor to generate an LU-factorization of matrix \mathbf{M}_{j-1} in a form equivalent to $\mathbf{L}_{j-1} \mathbf{L}_{j-1}^T$ wherein \mathbf{L}_{j-1} is a low-triangular matrix and \mathbf{L}_{j-1}^T is the transpose of \mathbf{L}_{j-1} ;

an eighth set of instructions that direct the data processor to generate a factorization of $\mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$ in a form equivalent to $\mathbf{T}_{j} \mathbf{T}_{j}^{T} = \mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$, where \mathbf{T}_{j}^{T} is the transpose of \mathbf{T}_{j} ; and

a ninth set of instructions that direct the data processor to generate an LUfactorization of matrix \mathbf{M}_j in a form equivalent to $\mathbf{L}_j \mathbf{L}_j^T$ from a plurality n of rank-one modifications of matrix L_{j-1} , each rank-one modification being based on a respective column of matrix T_j , where n is the number of rows in matrix M_j .

- 30. The computer program product of Claim 29 wherein the eighth set of instructions directs the data processor to generate matrix T_j from a Cholesky factorization of $G^T P_j G$.
- 31. The computer program product of Claim 29 wherein weighting matrix \mathbf{R}_j has a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein \mathbf{R}^{γ} and \mathbf{R}^{φ} are related to a common weighting matrix W and scaling

parameters σ_{γ} and σ_{φ} as follows $\left(\mathbf{R}^{\gamma}\right)^{-1} = \frac{1}{\sigma_{\gamma}^{2}}\mathbf{W}$, and $\left(\mathbf{R}^{\varphi}\right)^{-1} = \frac{1}{\sigma_{\varphi}^{2}}\mathbf{W}$; and

wherein the eighth set of instructions comprises:

instructions that direct the data processor to generate a scalar b in a form

equivalent to: $b = \frac{\sigma_{\gamma}^2 \lambda_{GPS}^2}{\sigma_{\gamma}^2 + \lambda_{GPS}^2 \sigma_{\varphi}^2}$, where λ_{GPS}^2 is the wavelength of the satellite signals,

instructions that direct the data processor to generate a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_j^{\gamma}$,

instructions that direct the data processor to generate a Householder matrix $\mathbf{S}_{\mathbf{H}\mathbf{H}}$ for matrix $\widetilde{\mathbf{H}}$, and

instructions that direct the data processor to generate matrix T_j in a form equivalent to:

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$$\mathbf{T}_{j} = \frac{1}{\sigma_{\varphi}} \mathbf{W}^{1/2} \mathbf{S}_{\mathbf{HH}} \begin{bmatrix} \sqrt{1 - \frac{b}{\lambda_{GPS}^{2}}} \mathbf{I}_{4} & | & \mathbf{O}_{4 \times (n-4)} \\ & --- & | & ---- \\ \mathbf{O}_{(n-4) \times 4} & | & \mathbf{I}_{(n-4) \times (n-4)} \end{bmatrix}.$$

32. The method of Claim 29 wherein weighting matrix \mathbf{R}_j is applied to a case where there is a first group of satellite signals having carrier signals in a first wavelength band and a second group of satellite signals having a carrier signals in a second wavelength band, the weighting frequency having a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein \mathbf{R}^{γ} and \mathbf{R}^{φ} are related to a common weighting matrix \mathbf{W} , the carrier wavelengths of the first group of signals as represented by matrix $\Lambda^{(1)}$, the carrier wavelengths of the second group of signals as represented by matrix $\Lambda^{(2)}$, the center wavelength of the first band as represented by λ_1 , the center wavelength of the first band as represented by λ_2 , and scaling parameters σ_{γ} and σ_{φ} , as follows:

$$\left(\mathbf{R}_{\mathbf{j}}^{\gamma}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} \end{bmatrix},$$

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$$\left(\mathbf{R}_{\mathbf{j}}^{\varphi}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(1)} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(2)} \end{bmatrix},$$

where
$$W^{(1)} = \frac{1}{\lambda_1^2} \Lambda^{(1)} W \Lambda^{(1)}$$
, $W^{(2)} = \frac{1}{\lambda_2^2} \Lambda^{(2)} W \Lambda^{(2)}$,

wherein step (h) of generating matrix T_j comprises the steps of:

generating a scalar
$$b$$
 in a form equivalent to:
$$b = \frac{\sigma_{\gamma}^2 \lambda_1^2 \lambda_2^2}{2\sigma_{\varphi}^2 \lambda_1^2 \lambda_2^2 + \sigma_{\gamma}^2 \lambda_1^2 + \sigma_{\gamma}^2 \lambda_2^2},$$

where λ_1 is the wavelength of a first group of satellite signals and λ_2 is the wavelength of a second group of satellite signals,

generating a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_{j}^{\gamma}$,

generating a Householder matrix S_{HH} for matrix $\widetilde{\mathbf{H}}$, and

generating matrix
$$\mathbf{T}_j$$
 in a form equivalent to: $\mathbf{T}_j = \frac{1}{\sigma_{\varphi}} \begin{bmatrix} \mathbf{A11} & | & \mathbf{O_{n \times n}} \\ --- & | & --- \\ \mathbf{A21} & | & \mathbf{A22} \end{bmatrix}$,

where sub-matrixces A11, A21, and A22 are as follows:

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A11 =
$$(W^{(1)})^{\frac{1}{2}}S_{HH}\begin{bmatrix} \sqrt{1-b/\lambda_1^2} I_4 & | & O_{4\times(n-4)} \\ ----- & | & ----- \\ O_{(n-4)\times 4} & | & I_{(n-4)} \end{bmatrix}$$
,

$$\mathbf{A21} = \left(\mathbf{W^{(2)}}\right)^{\frac{1}{2}} \mathbf{S_{HH}} \begin{bmatrix} -\frac{b}{\lambda_1 \lambda_2 \sqrt{1 - b/\lambda_1^2}} \mathbf{I_4} & | & \mathbf{O_{4 \times (n-4)}} \\ ----- & | & ----- \\ \mathbf{O_{(n-4) \times 4}} & | & \mathbf{O_{(n-4) \times (n-4)}} \end{bmatrix}, \text{ and}$$

$$\mathbf{A22} = \left(\mathbf{W^{(2)}}\right)^{\frac{1}{2}} \mathbf{S_{HH}} \begin{bmatrix} \sqrt{\frac{1 - b/\lambda_1^2 - b/\lambda_2^2}{1 - b/\lambda_1^2}} \ \mathbf{I_4} & | & \mathbf{O_{4 \times (n-4)}} \\ ---- & | & ----- \\ \mathbf{O_{(n-4) \times 4}} & | & \mathbf{I_{(n-4)}} \end{bmatrix}.$$

33. A computer program product for directing a data processor to estimate a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance, wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock

for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, the process receiving, for a plurality of two or more time moments j, the following inputs:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

a vector $\boldsymbol{\varphi_j}^{R}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),

a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, the computer program product comprising:

a computer-readable medium;

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a first set of instructions embodied on the computer-readable medium which directs the data processor to generate, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_i^R - \gamma_i^B) - (D_i^R - D_i^B)$;

a second set of instructions embodied on the computer-readable medium which directs the data processor to generate, for each time moment j, a vector $\Delta \varphi_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \varphi_j = (\varphi_j^R - \varphi_j^B) - \Lambda^{-1} \cdot (D_j^R - D_j^B)$, where Λ^{-1} is a diagonal matrix comprising the inverse wavelengths of the satellites;

a third set of instructions embodied on the computer-readable medium which directs the data processor to generate an LU-factorization of a matrix \mathbf{M} or a matrix inverse of matrix \mathbf{M} , the matrix \mathbf{M} being a function of at least $\boldsymbol{\Lambda}^{-1}$ and $\boldsymbol{H}_k^{\gamma}$, for index k of $\boldsymbol{H}_k^{\gamma}$ covering at least two of the time moments j; and

- a fourth set of instructions embodied on the computer-readable medium which directs the data processor to generate a vector \mathbf{N} of estimated floating ambiguities as a function of at least the set of range residuals $\Delta \gamma_k$, the set of phase residuals $\Delta \varphi_k$, and the LU-factorization of matrix \mathbf{M} or the matrix inverse of matrix \mathbf{M} .
- 34. The computer program product of Claim 33 wherein the third set of instructions directs the data processor to generate matrix **M** in a form equivalent to the summation

$$\left[\sum_{k=1}^{j} (\mathbf{G}^{\mathbf{T}} \mathbf{P}_{k} \mathbf{G})\right],$$

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- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix P_k has 2n rows, 2n columns, and a form which comprises a matrix equivalent to P_k = R_k⁻¹ R_k⁻¹ Q_k (Q_k^T R_k⁻¹ Q_k + qS_k)⁻¹ Q_k^T R_k⁻¹, where the matrix R_k is a weighting matrix, where the matrix R_k⁻¹ comprises an inverse of matrix R_k, where the matrix Q_k has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_k comprising matrix H_k^γ and the other of the sub-matrices of Q_k
 comprising the matrix product Λ⁻¹H_k^γ, and wherein the matrix Q_k^T comprises the transpose of matrix Q_k, and where the quantity qS_k is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

35. The computer program product of Claim 34 wherein the fourth set of instructions directs the data processor to generate matrix N in a form equivalent to:

$$\mathbf{N} = \mathbf{M}^{-1} \times \left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{\mu}_{k} + q \mathbf{g}_{k}) \right],$$

where the matrix \mathbf{M}^{-1} comprises an inverse of matrix of matrix \mathbf{M} , where the vector $\boldsymbol{\mu}_k$ comprises the vector $[\Delta \gamma_k, \Delta \varphi_k]^T$, and where the quantity $q\mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

36. The computer program product of Claim 35 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein the third set of instructions directs the data processor to generate matrix S_k for the k-th time moment in a form equivalent to:

$$\mathbf{S}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3 \times 1} \\ \mathbf{O}_{1 \times 3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{k}\|} \mathbf{r}_{k} \mathbf{r}_{k}^{T}$$

where \mathbf{r}_k is a vector comprising estimates of the three coordinates of the baseline vector at the k-th time moment, and a zero as fourth component, where \mathbf{r}_k^T is the vector transpose of \mathbf{r}_k , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_k for the k-th time moment in a form equivalent to:

$$\mathbf{g}_k = \mathbf{G}^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k (\mathbf{Q}_k^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k + q \mathbf{S}_k)^{-1} \mathbf{h}_k,$$

where:

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$$\mathbf{h}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \mathbf{r}_{k}.$$

37. An apparatus for estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance,

wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said apparatus receiving, for a plurality of two or more time moments j, the following inputs:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

a vector φ_j^R representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),

a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset,

said apparatus comprising:

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- (a) means for generating, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R \gamma_j^B) (D_j^R D_j^B)$;
 - (b) means for generating, for each time moment j, a vector $\Delta \boldsymbol{\varphi}_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \boldsymbol{\varphi}_j = (\boldsymbol{\varphi}_j^R \boldsymbol{\varphi}_j^B) \boldsymbol{\Lambda}^{-1} \cdot (\boldsymbol{D}_j^R \boldsymbol{D}_j^B)$, where $\boldsymbol{\Lambda}^{-1}$ is a diagonal matrix comprising the inverse wavelengths of the satellites;

- (c) means for generating, for time moment j = 1, an LU-factorization of a matrix \mathbf{M}_1 or a matrix inverse of matrix \mathbf{M}_1 , the matrix \mathbf{M}_1 being a function of at least $\boldsymbol{\Lambda}^{-1}$ and $\boldsymbol{H}_1^{\gamma}$;
- (d) means for generating, for time moment j = 1, a vector \mathbf{N}_1 as a function of at least $\Delta \gamma_I$, $\Delta \varphi_I$, and the LU-factorization of matrix \mathbf{M}_1 or the matrix inverse of matrix \mathbf{M}_1 ;
 - (e) means for generating, for an additional time moment $j \neq 1$, an LU-factorization of a matrix \mathbf{M}_j or a matrix inverse of matrix \mathbf{M}_j , the matrix \mathbf{M}_j being a function of at least $\mathbf{\Lambda}^{-1}$ and \mathbf{H}_j^{γ} ; and
 - (f) means for generating, for an additional time moment $j \neq 1$, a vector \mathbf{N}_j as a function of at least $\Delta \gamma_j$, $\Delta \varphi_j$, and the LU-factorization or matrix \mathbf{M}_j or the matrix inverse of matrix \mathbf{M}_j , the vector \mathbf{N}_j having estimates of the floating ambiguities.
 - 38. The apparatus of Claim 37 wherein means (c) comprises means for generating an LU-factorization for a matrix comprising a form equivalent to (G^TP_1G) , where:
 - the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and

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the matrix P₁ has 2n rows, 2n columns, and a form which comprises a matrix equivalent to (R₁⁻¹ - R₁⁻¹ Q₁(Q₁^T R₁⁻¹ Q₁ + qS₁)⁻¹Q₁^T R₁⁻¹) where the matrix R₁ is a weighting matrix, where the matrix R₁⁻¹ comprises an inverse of matrix R₁, where the matrix Q₁ has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q₁ comprising matrix H₁^γ and the other of the sub-matrices of Q₁ comprising the matrix product Λ⁻¹H₁^γ, and wherein the matrix Q₁^T comprises the transpose of matrix Q₁, and where the quantity qS_k is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may
be a non-zero weighting parameter and S_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

- 39. The apparatus of Claim 38 wherein means (d) comprises means for generating vector \mathbf{N}_1 to comprise a vector having a form equivalent to $\mathbf{M}_1^{-1} \left(\mathbf{G}^T \mathbf{P}_1 \ \boldsymbol{\mu}_1 + q \mathbf{g}_1 \right)$, where the matrix \mathbf{M}_1^{-1} comprises an inverse of matrix of matrix \mathbf{M}_1 , and where the vector $\boldsymbol{\mu}_1$ comprises the vector $[\Delta \boldsymbol{\gamma}_1, \Delta \boldsymbol{\varphi}_1]^T$, and where the quantity $q \mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.
- 40. The apparatus of Claim 37 wherein means (e) comprises means for generating an LU-factorization for a matrix comprising a form equivalent to

$$\mathbf{M}_{j} = \mathbf{M}_{j-1} + \mathbf{G}^{T} \mathbf{P}_{j} \mathbf{G}$$
, where:

- \mathbf{M}_{j-1} comprises the matrix \mathbf{M}_1 of generated by means (c) when j = 2 and comprises the matrix \mathbf{M}_j generated by means (e) for the j-1 time moment when j > 2,
- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprising an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix P_j has 2n rows, 2n columns, and a form which comprise a matrix equivalent to (R_j⁻¹ R_j⁻¹ Q_j(Q_j^T R_j⁻¹ Q_j + qS_j)⁻¹Q_j^T R_j⁻¹) where the matrix R_j is a weighting matrix, where the matrix R_j⁻¹ comprises an inverse of matrix R_j, where the matrix Q_j has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_j comprising matrix H_j^γ and the other of the sub-matrices of Q_j
 comprising the matrix product Λ⁻¹H_j^γ, and wherein the matrix Q_j^T comprises the transpose of matrix Q_j, and where the quantity qS_j is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_j may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

41. The apparatus of Claim 40 wherein means (f) comprises means for generating vector N_j to comprise a vector having a form equivalent to

$$\mathbf{N}_{j-1} + \mathbf{M}_{j}^{-1} \left[\mathbf{G}^{T} \mathbf{P}_{j} \left(\mathbf{\mu}_{j} - \mathbf{G} \mathbf{N}_{j-1} \right) + q \mathbf{g}_{j} \right]$$
, where the matrix \mathbf{M}_{j}^{-1} comprises an inverse of matrix of matrix \mathbf{M}_{j} , where the vector $\mathbf{\mu}_{j}$ comprises the vector $\left[\Delta \gamma_{j}, \Delta \varphi_{j} \right]^{T}$, and where the vector \mathbf{N}_{j-1} comprises the vector \mathbf{N}_{1} generated by means (d) when $j = 2$ and comprises the vector \mathbf{N}_{j-1} generated by means (f) for the j -1 time moment when $j > 2$, and where the quantity $q\mathbf{g}_{j}$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_{j} may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

42. The apparatus of claim 39 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein means (c) comprises means for generating matrix S_1 in a form equivalent to:

$$\mathbf{S}_{1} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{1}\|} \mathbf{r}_{1} \mathbf{r}_{1}^{T}$$

where \mathbf{r}_1 is a vector comprising estimates of the three coordinates of the baseline vector for the time moment j=1 and a zero as fourth component, where \mathbf{r}_1^T is the vector transpose of \mathbf{r}_1 , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein means (d) comprises means for generating vector \mathbf{g}_1 for the time moment j=1 in a form equivalent to:

$$\mathbf{g}_1 = \mathbf{G}^T \mathbf{R}_1^{-1} \mathbf{Q}_1 (\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1)^{-1} \mathbf{h}_1,$$

where:

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$$\mathbf{h}_1 = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_1\|}\right) \mathbf{r}_1.$$

43. The apparatus of Claim 41 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein means (e) comprises means for generating matrix S_i in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right)\left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} \quad 0\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\mathbf{r}j\mathbf{r}j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein means (f) comprises means for generating vector \mathbf{g}_j for the j-th time moment in a form equivalent to:

$$\mathbf{g}_{j} = \mathbf{G}^{\mathsf{T}} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} (\mathbf{Q}_{j}^{\mathsf{T}} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} + q \mathbf{S}_{j})^{-1} \mathbf{h}_{j},$$

where:

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$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

- 44. The apparatus of Claim 40 wherein means (e) for generating the LU-factorization of M_i comprises:
- (g) means for generating an LU-factorization of matrix \mathbf{M}_{j-I} in a form equivalent to $\mathbf{L}_{j-1} \mathbf{L}_{j-1}^T$ wherein \mathbf{L}_{j-1} is a low-triangular matrix and \mathbf{L}_{j-1}^T is the
- 5 transpose of L_{i-1} ;
 - (h) means for generating a factorization of $\mathbf{G}^{\mathbf{T}} \mathbf{P}_{j} \mathbf{G}$ in a form equivalent to $\mathbf{T}_{j} \mathbf{T}_{j}^{\mathbf{T}} = \mathbf{G}^{\mathbf{T}} \mathbf{P}_{i} \mathbf{G}$, where $\mathbf{T}_{i}^{\mathbf{T}}$ is the transpose of \mathbf{T}_{i} ; and
 - (i) means for generating an LU-factorization of matrix \mathbf{M}_j in a form equivalent to $\mathbf{L}_j \mathbf{L}_j^T$ from a plurality n of rank-one modifications of matrix \mathbf{L}_{j-1} , each rank-one modification being based on a respective column of matrix \mathbf{T}_j , where n is the number of rows in matrix \mathbf{M}_j .
 - 45. The apparatus of Claim 44 wherein means (h) generates matrix \mathbf{T}_j from a Cholesky factorization of $\mathbf{G}^T \mathbf{P}_j \mathbf{G}$.

46. The apparatus of Claim 44 wherein weighting matrix \mathbf{R}_j has a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein \mathbf{R}^{γ} and \mathbf{R}^{φ} are related to a common weighting matrix \mathbf{W} and scaling

parameters σ_{γ} and σ_{φ} as follows $\left(\mathbf{R}^{\gamma}\right)^{-1} = \frac{1}{\sigma_{\gamma}^{2}}\mathbf{W}$, and $\left(\mathbf{R}^{\varphi}\right)^{-1} = \frac{1}{\sigma_{\varphi}^{2}}\mathbf{W}$,

wherein means (h) of generating matrix T_i comprises:

means for generating a scalar b in a form equivalent to: $b = \frac{\sigma_{\gamma}^2 \lambda_{GPS}^2}{\sigma_{\gamma}^2 + \lambda_{GPS}^2 \sigma_{\varphi}^2}$, where

 λ_{GPS}^2 is the wavelength of the satellite signals,

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means for generating a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_{j}^{\gamma}$,

means for generating a Householder matrix \mathbf{S}_{HH} for matrix $\widetilde{\mathbf{H}}$, and means for generating matrix \mathbf{T}_j in a form equivalent to:

$$\mathbf{T}_{j} = \frac{1}{\sigma_{\varphi}} \mathbf{W}^{1/2} \mathbf{S}_{\mathbf{HH}} \begin{bmatrix} \sqrt{\left(1 - \frac{b}{\lambda_{GPS}^{2}}\right)} \mathbf{I}_{4} & | \mathbf{O}_{4 \times (n-4)} \\ --- & | ---- \\ \mathbf{O}_{(n-4) \times 4} & | \mathbf{I}_{(n-4) \times (n-4)} \end{bmatrix}.$$

47. The method of Claim 44 wherein weighting matrix \mathbf{R}_j is applied to a case where there is a first group of satellite signals having carrier signals in a first wavelength band and a second group of satellite signals having a carrier signals in a second wavelength band, the weighting frequency having a form equivalent to:

$$R_{j} = \begin{bmatrix} R_{j}^{\gamma} & 0 \\ 0 & R_{j}^{\varphi} \end{bmatrix}$$
, where \mathbf{R}^{γ} and \mathbf{R}^{φ} are weighting matrices;

wherein ${\bf R}^{\gamma}$ and ${\bf R}^{\varphi}$ are related to a common weighting matrix ${\bf W}$, the carrier wavelengths of the first group of signals as represented by matrix ${\bf \Lambda}^{(1)}$, the carrier

wavelengths of the second group of signals as represented by matrix $\Lambda^{(2)}$, the center wavelength of the first band as represented by λ_1 , the center wavelength of the first band as represented by λ_2 , and scaling parameters σ_{γ} and σ_{φ} , as follows:

$$\left(\mathbf{R}_{\mathbf{j}}^{\gamma}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\gamma}^{2}} \mathbf{W} \end{bmatrix},$$

$$\left(\mathbf{R}_{\mathbf{j}}^{\varphi}\right)^{-1} = \begin{bmatrix} \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(1)} & | & \mathbf{O}_{n \times n} \\ --- & | & --- \\ \mathbf{O}_{n \times n} & | & \frac{1}{\sigma_{\varphi}^{2}} \mathbf{W}^{(2)} \end{bmatrix},$$

where
$$W^{(1)} = \frac{1}{\lambda_1^2} \Lambda^{(1)} W \Lambda^{(1)}$$
, $W^{(2)} = \frac{1}{\lambda_2^2} \Lambda^{(2)} W \Lambda^{(2)}$,

wherein step (h) of generating matrix T_i comprises the steps of:

generating a scalar
$$b$$
 in a form equivalent to:
$$b = \frac{\sigma_{\gamma}^2 \lambda_1^2 \lambda_2^2}{2\sigma_{\varphi}^2 \lambda_1^2 \lambda_2^2 + \sigma_{\gamma}^2 \lambda_1^2 + \sigma_{\gamma}^2 \lambda_2^2},$$

where λ_1 is the wavelength of a first group of satellite signals and λ_2 is the wavelength of a second group of satellite signals,

generating a matrix $\widetilde{\mathbf{H}}$ in a form equivalent to $\widetilde{\mathbf{H}} = \mathbf{W}^{1/2} \mathbf{H}_{i}^{\gamma}$,

generating a Householder matrix S_{HH} for matrix \widetilde{H} , and

generating matrix
$$\mathbf{T}_j$$
 in a form equivalent to: $\mathbf{T}_j = \frac{1}{\sigma_{\varphi}} \begin{bmatrix} \mathbf{A11} & | & \mathbf{O}_{\mathbf{n} \times \mathbf{n}} \\ --- & | & --- \\ \mathbf{A21} & | & \mathbf{A22} \end{bmatrix}$,

where sub-matrixces A11, A21, and A22 are as follows:

$$A11 = \left(W^{(1)}\right)^{\frac{1}{2}} S_{HH} \begin{bmatrix} \sqrt{1 - b/\lambda_{1}^{2}} & I_{4} & | & O_{4\times(n-4)} \\ ----- & | & ----- \\ O_{(n-4)\times 4} & | & I_{(n-4)} \end{bmatrix},$$

$$A21 = \left(W^{(2)}\right)^{\frac{1}{2}} S_{HH} \begin{bmatrix} -\frac{b}{\lambda_{1}\lambda_{2}\sqrt{1 - b/\lambda_{1}^{2}}} & I_{4} & | & O_{4\times(n-4)} \\ ----- & | & ----- \\ O_{(n-4)\times 4} & | & O_{(n-4)\times(n-4)} \end{bmatrix}, \text{ and }$$

$$A22 = \left(W^{(2)}\right)^{\frac{1}{2}} S_{HH} \begin{bmatrix} \sqrt{\frac{1 - b/\lambda_{1}^{2} - b/\lambda_{2}^{2}}{1 - b/\lambda_{1}^{2}}} & I_{4} & | & O_{4\times(n-4)} \\ ----- & | & ----- \\ O_{(n-4)\times 4} & | & I_{(n-4)} \end{bmatrix}.$$

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48. An apparatus for estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance, wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said apparatus receiving, for a plurality of two or more time moments j, the following inputs:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector γ_j^R representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

- a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),
- a vector $\boldsymbol{\varphi_j}^{\mathbf{R}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),
- a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset,

said apparatus comprising:

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- (a) means for generating, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R \gamma_j^B) (D_j^R D_j^B)$, said means generating a set of range residuals $\Delta \gamma_k$, k=1,...,j;
- (b) means for generating, for each time moment j, a vector $\Delta \boldsymbol{\varphi}_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \boldsymbol{\varphi}_j = (\boldsymbol{\varphi}_j^{\mathbf{R}} \boldsymbol{\varphi}_j^{\mathbf{B}}) \boldsymbol{\Lambda}^{-1} \cdot (\boldsymbol{D}_j^{\mathbf{R}} \boldsymbol{D}_j^{\mathbf{B}})$, where $\boldsymbol{\Lambda}^{-1}$ is a diagonal matrix comprising the inverse wavelengths of the satellites, said means generating a set of phase residuals $\Delta \boldsymbol{\varphi}_k$, k=1,...,j;
 - (c) means for generating an LU-factorization of a matrix \mathbf{M} or a matrix inverse of matrix \mathbf{M} , the matrix \mathbf{M} being a function of at least $\boldsymbol{\Lambda}^{-1}$ and $\boldsymbol{H}_{k}^{\gamma}$, for index k of $\boldsymbol{H}_{k}^{\gamma}$ covering at least two of the time moments j;
 - (d) means for generating a vector \mathbf{N} of estimated floating ambiguities as a function of at least the set of range residuals $\Delta \gamma_k$, the set of phase residuals $\Delta \varphi_k$, and the LU-factorization of matrix \mathbf{M} or the matrix inverse of matrix \mathbf{M} .
 - 49. The apparatus of Claim 48 wherein means (c) comprises means for generating matrix \mathbf{M} in a form equivalent to the summation $\left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{G})\right]$, where:

- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and

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- the matrix P_k has 2n rows, 2n columns, and a form which comprises a matrix equivalent to P_k = R_k⁻¹ R_k⁻¹ Q_k (Q_k^T R_k⁻¹ Q_k + qS_k)⁻¹ Q_k^T R_k⁻¹, where the matrix R_k is a weighting matrix, where the matrix R_k⁻¹ comprises an inverse of matrix R_k, where the matrix Q_k has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_k comprising matrix H_k^γ and the other of the sub-matrices of Q_k comprising the matrix product Λ⁻¹H_k^γ, and wherein the matrix Q_k^T comprises the transpose of matrix Q_k, and where the quantity qS_k is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.
 - 50. The apparatus of Claim 49 wherein means (d) comprises means for generating matrix N in a form equivalent to:

$$\mathbf{N} = \mathbf{M}^{-1} \times \left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{\mu}_{k} + q \mathbf{g}_{k}) \right],$$

where the matrix \mathbf{M}^{-1} comprises an inverse of matrix of matrix \mathbf{M} , where the vector $\boldsymbol{\mu}_k$ comprises the vector $[\Delta \gamma_k, \Delta \varphi_k]^T$, and where the quantity $q\mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

51. The apparatus of claim 50 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein means (c) comprises means for generating matrix S_k for the k-th time moment in a form equivalent to:

$$\mathbf{S}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{k}\|} \mathbf{r}_{k} \mathbf{r}_{k}^{T}$$

where \mathbf{r}_k is a vector comprising estimates of the three coordinates of the baseline vector at the k-th time moment, and a zero as fourth component, where \mathbf{r}_k^T is the vector transpose of \mathbf{r}_k , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein means (d) generates vector \mathbf{g}_k for the k-th time moment in a form equivalent to:

$$\mathbf{g}_k = \mathbf{G}^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k (\mathbf{Q}_k^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k + q \mathbf{S}_k)^{-1} \mathbf{h}_k,$$

where:

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$$\mathbf{h}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \mathbf{r}_{k}.$$

52. (Claims for Japan) A computer program to be installed in a computer for controlling the computer to perform the process of estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R) separated by a distance, wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said method receiving, for a plurality of two or more time moments j, the following inputs:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector \mathbf{y}_{j}^{R} representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

a vector $\boldsymbol{\varphi_i}^{\mathbf{R}}$ representative of a plurality of full phase measurements of the 20 satellite carrier signals measured by the second navigation receiver (R),

a geometric Jacobian matrix H_i^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, said process comprising:

- (a) generating, for each time moment j, a vector $\Delta \gamma_i$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation $\Delta \gamma_i = (\gamma_i^R - \gamma_i^B) - (D_i^R - D_i^B);$ receivers in the form of:
- (b) generating, for each time moment j, a vector $\Delta \boldsymbol{\varphi}_i$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of:

$$\Delta \varphi_j = (\varphi_j^R - \varphi_j^B) - \Lambda^{-1} \cdot (D_j^R - D_j^B),$$

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 $\Delta \varphi_j = (\varphi_j^R - \varphi_j^B) - \Lambda^{-1} \cdot (D_j^R - D_j^B),$ where Λ^{-1} is a diagonal matrix comprising the inverse wavelengths of the satellites;

- (c) generating, for time moment j = 1, an LU-factorization of a matrix M_1 or a matrix inverse of matrix M_1 , the matrix M_1 being a function of at least Λ^{-1} and H_1^{γ} ;
- (d) generating, for time moment j = 1, a vector N_1 as a function of at least $\Delta \gamma_I$, $\Delta \varphi_1$, and the LU-factorization of matrix M_1 or the matrix inverse of matrix M_1 ;
- (e) generating, for an additional time moment $i \neq 1$, an LU-factorization of a matrix \mathbf{M}_j or a matrix inverse of matrix \mathbf{M}_j , the matrix \mathbf{M}_j being a function of at least $\boldsymbol{\Lambda}^{-1}$, H_j^{γ} and an instance of matrix M generated for a different time moment; and
- (f) generating, for an additional time moment $j \neq 1$, a vector N_i as a function of at least $\Delta \gamma_j$, $\Delta \varphi_j$, and the LU-factorization or matrix M_j or the matrix inverse of matrix M_j , the vector N_i having estimates of the floating ambiguities.
- 53. The computer program of Claim 52 wherein step (c) of the process comprises generating an LU-factorization for a matrix comprising a form equivalent to $(\mathbf{G}^{\mathbf{T}}\mathbf{P}_{1}\mathbf{G})$, where:

- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix,
 one of the sub-matrices comprises an n x n zero matrix and the other sub-matrix comprising an n x n identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and
 - the matrix \mathbf{P}_1 has 2n rows, 2n columns, and a form which comprises a matrix equivalent to $\left(\mathbf{R}_1^{-1} \mathbf{R}_1^{-1} \mathbf{Q}_1 \left(\mathbf{Q}_1^{\mathbf{T}} \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1\right)^{-1} \mathbf{Q}_1^{\mathbf{T}} \mathbf{R}_1^{-1}\right)$ where the matrix \mathbf{R}_1
- is a weighting matrix, where the matrix \mathbf{R}_1^{-1} comprises an inverse of matrix \mathbf{R}_1 , where the matrix \mathbf{Q}_1 has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of \mathbf{Q}_1 comprising matrix \mathbf{H}_I^{γ} and the other of the sub-matrices of \mathbf{Q}_1 comprising the matrix product $\mathbf{\Lambda}^{-1}\mathbf{H}_I^{\gamma}$, and wherein the matrix $\mathbf{Q}_1^{\mathsf{T}}$ comprises the transpose of matrix \mathbf{Q}_1 , and where the quantity $\mathbf{q}\mathbf{S}_k$ is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where \mathbf{q} may be a non-zero weighting parameter and \mathbf{S}_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.
 - 54. The computer program of Claim 53 wherein step (d) of the process comprises the step of generating vector \mathbf{N}_1 to comprise a vector having a form equivalent to $\mathbf{M}_1^{-1} \left(\mathbf{G}^T \mathbf{P}_1 \ \boldsymbol{\mu}_1 + q \mathbf{g}_1 \right)$, where the matrix \mathbf{M}_1^{-1} comprises an inverse of matrix of matrix \mathbf{M}_1 , and where the vector $\boldsymbol{\mu}_1$ comprises the vector $[\Delta \boldsymbol{\gamma}_1, \Delta \boldsymbol{\varphi}_1]^T$, and where the quantity $q \mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.
 - 55. The computer program of Claim 52 wherein step (e) of the process comprises generating an LU-factorization for a matrix comprising a form equivalent to

$$\mathbf{M}_j = \mathbf{M}_{j-1} + \mathbf{G}^{\mathrm{T}} \mathbf{P}_j \mathbf{G}$$
, where:

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• \mathbf{M}_{j-1} comprises the matrix \mathbf{M}_1 of step (c) when j=2 and comprises the matrix \mathbf{M}_j of step (e) for the j-1 time moment when j>2,

- the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprising an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
- the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix P_j has 2n rows, 2n columns, and a form which comprise a matrix equivalent to (R_j⁻¹ R_j⁻¹ Q_j(Q_j^T R_j⁻¹ Q_j + qS_j)⁻¹Q_j^T R_j⁻¹) where the matrix R_j, where the matrix Q_j has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of Q_j comprising matrix H_j^γ and the other of the sub-matrices of Q_j
 comprising the matrix product Λ⁻¹H_j^γ, and wherein the matrix Q_j^T comprises the transpose of matrix Q_j, and where the quantity qS_j is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and S_j may be a non-zero matrix when the distance

56. The computer program of Claim 55 wherein step (f) of the process comprises the step of generating vector \mathbf{N}_{j} to comprise a vector having a form equivalent to $\mathbf{N}_{j-1} + \mathbf{M}_{j}^{-1} \left| \mathbf{G}^{\mathbf{T}} \mathbf{P}_{j} \left(\mathbf{\mu}_{j} - \mathbf{G} \mathbf{N}_{j-1} \right) + q \mathbf{g}_{j} \right|, \text{ where the matrix } \mathbf{M}_{j}^{-1} \text{ comprises an}$

between the first and second navigation receivers is constrained.

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inverse of matrix of matrix \mathbf{M}_j , where the vector $\boldsymbol{\mu}_j$ comprises the vector $[\Delta \gamma_j, \Delta \boldsymbol{\varphi}_j]^T$, and where the vector \mathbf{N}_{j-1} comprises the vector \mathbf{N}_1 generated by step (d) when j=2 and comprises the vector \mathbf{N}_{j-1} generated by step (f) for the j-1 time moment when j>2, and where the quantity $q\mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_j may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

57. The computer program of claim 56 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) of the process generates matrix S_1 in a form equivalent to:

$$\mathbf{S}_{1} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{1}\|} \mathbf{r}_{1} \mathbf{r}_{1}^{T}$$

where \mathbf{r}_1 is a vector comprising estimates of the three coordinates of the baseline vector for the time moment j=1 and a zero as fourth component, where \mathbf{r}_1^T is the vector transpose of \mathbf{r}_1 , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_1 for the time moment j=1 in a form equivalent

$$\mathbf{g}_1 = \mathbf{G}^T \mathbf{R}_1^{-1} \mathbf{Q}_1 (\mathbf{Q}_1^T \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1)^{-1} \mathbf{h}_1$$

where:

to:

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$$\mathbf{h}_1 = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_1\|}\right) \mathbf{r}_1.$$

58. The computer program of claim 56 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (e) of the process generates matrix S_i in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right)\left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1}\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\mathbf{r}j\mathbf{r}j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (f) generates vector \mathbf{g}_j for the j-th time moment in a form equivalent

10 to:

$$\mathbf{g}_{i} = \mathbf{G}^{T} \mathbf{R}_{i}^{-1} \mathbf{Q}_{i} (\mathbf{Q}_{i}^{T} \mathbf{R}_{i}^{-1} \mathbf{Q}_{i} + q \mathbf{S}_{i})^{-1} \mathbf{h}_{i},$$

where:

$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

- 59. The computer program of Claim 55 wherein the generation of the LU-factorization in step (e) of the process comprises the steps of:
- (g) generating an LU-factorization of matrix \mathbf{M}_{j-1} in a form equivalent to $\mathbf{L}_{j-1} \mathbf{L}_{j-1}^T$ wherein \mathbf{L}_{j-1} is a low-triangular matrix and \mathbf{L}_{j-1}^T is the transpose of \mathbf{L}_{j-1} ;

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- (h) generating a factorization of $\mathbf{G}^{\mathsf{T}} \mathbf{P}_{j} \mathbf{G}$ in a form equivalent to $\mathbf{T}_{j} \mathbf{T}_{j}^{\mathsf{T}} = \mathbf{G}^{\mathsf{T}} \mathbf{P}_{j} \mathbf{G}$, where $\mathbf{T}_{i}^{\mathsf{T}}$ is the transpose of \mathbf{T}_{i} ;
- (i) generating an LU-factorization of matrix \mathbf{M}_j in a form equivalent to $\mathbf{L}_j \mathbf{L}_j^T$ from a plurality n of rank-one modifications of matrix \mathbf{L}_{j-1} , each rank-one modification being based on a respective column of matrix \mathbf{T}_i , where n is the number of rows in matrix \mathbf{M}_i .
- 60. The computer program of Claim 59 wherein step (h) of the process generates matrix \mathbf{T}_i from a Cholesky factorization of $\mathbf{G}^T \mathbf{P}_i \mathbf{G}$.
- 61. The computer program of Claim 55 wherein step (d) of the process comprises generating a vector \mathbf{B}_1 to comprise a vector having a form equivalent to $\mathbf{G}^T \mathbf{P}_1 \mu_1 + q \mathbf{g}_1$, where the vector $\boldsymbol{\mu}_1$ comprises the vector $[\boldsymbol{\Delta}\boldsymbol{\gamma}_1, \boldsymbol{\Delta}\boldsymbol{\varphi}_1]^T$, and where the quantity $q\mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_j may be a non-zero vector when the distance between the first and second navigation receivers is constrained; and

wherein step (f) of the process further comprises generating, for each time moment $j \neq 1$, a vector \mathbf{B}_j to comprise a matrix having a form equivalent to $\mathbf{B}_{j-1} + \mathbf{G}^T \mathbf{P}_j \mu_j + q\mathbf{g}_j$, where the vector μ_j comprises the vector $[\Delta \gamma_j, \Delta \varphi_j]^T$, and where the vector \mathbf{B}_{j-1} is the vector \mathbf{B}_1 generated by step (d) when j = 2 and comprises the vector generated by step (f) for the for the j-1 time moment when j > 2, and where the quantity $q\mathbf{g}_j$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_j may be a non-zero vector when the distance between the first and second navigation receivers is constrained; and

wherein step (f) of the process further comprises generating vector \mathbf{N}_j to comprise a vector having a form equivalent to $\mathbf{N}_j = [\mathbf{M}_j]^{-1} \mathbf{B}_j$, where the matrix \mathbf{M}_j^{-1} comprises an inverse of matrix of matrix \mathbf{M}_j .

62. The computer program of claim 61 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) of the process generates matrix S_1 in a form equivalent to:

$$\mathbf{S}_{1} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\right)\left(\begin{matrix}\mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0\end{matrix}\right) + \frac{L_{RB}}{\|\mathbf{r}_{1}\|}\mathbf{r}_{1}\mathbf{r}_{1}^{T}$$

where \mathbf{r}_1 is a vector comprising estimates of the three coordinates of the baseline vector for the time moment j=1 and a zero as fourth component, where \mathbf{r}_1^T is the vector transpose of \mathbf{r}_1 , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) generates vector \mathbf{g}_1 for the time moment j=1 in a form equivalent

10 to:

$$\mathbf{g}_1 = \mathbf{G}^{\mathsf{T}} \mathbf{R}_1^{-1} \mathbf{Q}_1 (\mathbf{Q}_1^{\mathsf{T}} \mathbf{R}_1^{-1} \mathbf{Q}_1 + q \mathbf{S}_1)^{-1} \mathbf{h}_1,$$

where:

$$\mathbf{h}_1 = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_1\|}\right) \mathbf{r}_1.$$

63. The computer program of claim 61 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (e) of the process generates matrix S_i in a form equivalent to:

$$\mathbf{S}j = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \left(\mathbf{I}_{3} \quad \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} \quad 0\right) + \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|} \mathbf{r} j \mathbf{r} j^{T}$$

where \mathbf{r}_j is a vector comprising estimates of the three coordinates of the baseline vector for the j-th time moment and a zero as fourth vector component, where \mathbf{r}_j^T is the vector transpose of \mathbf{r}_j , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (f) of the process generates vector \mathbf{g}_j for the j-th time moment in a form equivalent to:

$$\mathbf{g}_{j} = \mathbf{G}^{\mathsf{T}} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} (\mathbf{Q}_{j}^{\mathsf{T}} \mathbf{R}_{j}^{-1} \mathbf{Q}_{j} + q \mathbf{S}_{j})^{-1} \mathbf{h}_{j},$$

where:

$$\mathbf{h}_{j} = \left(1 - \frac{L_{RB}}{\left\|\mathbf{r}_{j}\right\|}\right) \mathbf{r}_{j}.$$

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64. (Claims for Japan) A computer program to be installed in a computer for controlling the computer to perform the process of estimating a set of floating ambiguities associated with a set of phase measurements of a plurality n of satellite carrier signals made by a first navigation receiver (B) and a second navigation receiver (R), wherein a baseline vector (x^o, y^o, z^o) relates the position of the second receiver to the first receiver, each satellite carrier signal being transmitted by a satellite and having a wavelength, wherein each receiver has a time clock for referencing its measurements and wherein any difference between the time clocks may be represented by an offset, said method receiving, for a plurality of two or more time moments j, the following inputs for each time moment j:

a vector γ_j^B representative of a plurality of pseudo-ranges measured by the first navigation receiver (B) and corresponding to the plurality of satellite carrier signals,

a vector \mathbf{y}_{j}^{R} representative of a plurality of pseudo-ranges measured by the second navigation receiver (R) and corresponding to the plurality of satellite carrier signals,

a vector D_j^B representative of a plurality of estimated distances between the satellites and the first navigation receiver (B),

a vector D_j^R representative of a plurality of estimated distances between the satellites and the second navigation receiver (R),

a vector $\boldsymbol{\varphi_j}^{\mathbf{B}}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the first navigation receiver (B),

a vector $\boldsymbol{\varphi_j}^{R}$ representative of a plurality of full phase measurements of the satellite carrier signals measured by the second navigation receiver (R),

a geometric Jacobian matrix H_j^{γ} whose matrix elements are representative of the changes in the distances between the satellites and one of the receivers that would be caused by changes in that receiver's position and time clock offset, the process comprising:

(a) generating, for each time moment j, a vector $\Delta \gamma_j$ of a plurality of range residuals of pseudo-range measurements made by the first and second navigation receivers in the form of: $\Delta \gamma_j = (\gamma_j^R - \gamma_j^B) - (D_j^R - D_j^B)$, said step generating a set of range residuals $\Delta \gamma_k$, k=1,...,j;

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- (b) generating, for each time moment j, a vector $\Delta \boldsymbol{\varphi}_j$ of a plurality of phase residuals of full phase measurements made by the first and second navigation receivers in the form of: $\Delta \boldsymbol{\varphi}_j = (\boldsymbol{\varphi}_j^R \boldsymbol{\varphi}_j^B) \boldsymbol{\Lambda}^{-1} \cdot (\boldsymbol{D}_j^R \boldsymbol{D}_j^B)$, where $\boldsymbol{\Lambda}^{-1}$ is a diagonal matrix comprising the inverse wavelengths of the satellites, said step generating a set of phase residuals $\Delta \boldsymbol{\varphi}_k$, k=1,...,j;
- (c) generating an LU-factorization of a matrix \mathbf{M} or a matrix inverse of matrix \mathbf{M} , the matrix \mathbf{M} being a function of at least $\boldsymbol{\Lambda}^{-1}$ and $\boldsymbol{H}_k^{\gamma}$, for index k of $\boldsymbol{H}_k^{\gamma}$ covering at least two of the time moments j;
- (d) generating a vector **N** of estimated floating ambiguities as a function of at least the set of range residuals $\Delta \gamma_k$, the set of phase residuals $\Delta \varphi_k$, and the LU-factorization of matrix **M** or the matrix inverse of matrix **M**.
 - 65. The computer program of Claim 64 wherein step (c) of the process comprises generating matrix \mathbf{M} in a form equivalent to the summation $\left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{G})\right]$, where:
 - the matrix G has 2n rows, n columns, an upper sub-matrix, and a lower sub-matrix, one of the sub-matrices comprises an $n \times n$ zero matrix and the other sub-matrix comprising an $n \times n$ identity matrix,
 - the matrix G^T comprises the transpose matrix of matrix G, and
- the matrix \mathbf{P}_k has 2n rows, 2n columns, and a form which comprises a matrix equivalent to $\mathbf{P}_k = \mathbf{R}_k^{-1} \mathbf{R}_k^{-1} \mathbf{Q}_k \left(\mathbf{Q}_k^T \mathbf{R}_k^{-1} \mathbf{Q}_k + q \mathbf{S}_k \right)^{-1} \mathbf{Q}_k^T \mathbf{R}_k^{-1}$, where the matrix \mathbf{R}_k is a weighting matrix, where the matrix \mathbf{R}_k^{-1} comprises an inverse of matrix \mathbf{R}_k , where the matrix \mathbf{Q}_k has an upper sub-matrix and a lower sub-matrix, one of the sub-matrices of \mathbf{Q}_k comprising matrix \mathbf{H}_k^{γ} and the other of the sub-matrices of \mathbf{Q}_k

comprising the matrix product $\Lambda^{-1}H_k^{\gamma}$, and wherein the matrix $\mathbf{Q}_k^{\mathrm{T}}$ comprises the transpose of matrix \mathbf{Q}_k , and where the quantity $q\mathbf{S}_k$ is a zero matrix when the distance between the first and second navigation receivers is unconstrained and where q may be a non-zero weighting parameter and \mathbf{S}_k may be a non-zero matrix when the distance between the first and second navigation receivers is constrained.

66. The computer program of Claim 65 wherein step (d) of the process comprises generating matrix N in a form equivalent to:

$$\mathbf{N} = \mathbf{M}^{-1} \times \left[\sum_{k=1}^{j} (\mathbf{G}^{T} \mathbf{P}_{k} \mathbf{\mu}_{k} + q \mathbf{g}_{k}) \right],$$

where the matrix \mathbf{M}^{-1} comprises an inverse of matrix of matrix \mathbf{M} , where the vector $\boldsymbol{\mu}_k$ comprises the vector $[\Delta \gamma_k, \Delta \varphi_k]^T$, and where the quantity $q\mathbf{g}_k$ is a zero vector when the distance between the first and second navigation receivers is unconstrained and where q may be non-zero and \mathbf{g}_k may be a non-zero vector when the distance between the first and second navigation receivers is constrained.

67. The computer program of claim 66 wherein the distance between the first and second navigation receivers is constrained to a distance L_{RB} , wherein step (c) of the process generates matrix S_k for the k-th time moment in a form equivalent to:

$$\mathbf{S}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \begin{pmatrix} \mathbf{I}_{3} & \mathbf{O}_{3\times1} \\ \mathbf{O}_{1\times3} & 0 \end{pmatrix} + \frac{L_{RB}}{\|\mathbf{r}_{k}\|} \mathbf{r}_{k} \mathbf{r}_{k}^{T}$$

where \mathbf{r}_k is a vector comprising estimates of the three coordinates of the baseline vector at the k-th time moment, and a zero as fourth component, where \mathbf{r}_k^T is the vector transpose of \mathbf{r}_k , where \mathbf{I}_3 is the 3-by-3 identity matrix, where \mathbf{O}_{1x3} is a row vector of three zeros, and where \mathbf{O}_{3x1} is a column vector of three zeros; and

wherein step (d) of the process generates vector \mathbf{g}_k for the k-th time moment in a form equivalent to:

$$\mathbf{g}_k = \mathbf{G}^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k (\mathbf{Q}_k^{\mathrm{T}} \mathbf{R}_k^{-1} \mathbf{Q}_k + q \mathbf{S}_k)^{-1} \mathbf{h}_k,$$

where:

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$$\mathbf{h}_{k} = \left(1 - \frac{L_{RB}}{\|\mathbf{r}_{k}\|}\right) \mathbf{r}_{k}.$$

68. The computer program of Claim 65 wherein at least one of the weighting matrices \mathbf{R}_k comprises an identity matrix multiplied by a scalar quantity.